

**Ae/AM/CE/ME 102c**  
**Mechanics of Structures and Solids – Spring 2007-08**  
**Assignment 3, Due Apr 24, 9:00AM, in class**

Instructor: G. Ravichandran 109 Firestone ravi@caltech.edu x4525

Office Hours:

Mike Silva 204 Firestone [michaels@caltech.edu](mailto:michaels@caltech.edu) Wed 1-3pm

Ling Zheng 212E Thomas [lzheng@caltech.edu](mailto:lzheng@caltech.edu) Mon 3-5pm

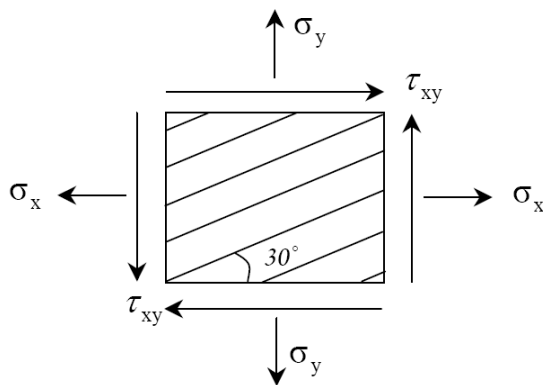
Grading TA: Ling Zheng

**\*\*\*Please indicate the number of hours spent on this assignment\*\*\***

**Problem 1 (10 points)**

A unidirectional glass fiber reinforced epoxy lamina with 60% fiber volume fraction is subjected to biaxial loading as shown in the figure, with  $\sigma_x=200$  MPa,  $\sigma_y=-80$  MPa,  $\tau_{xy}=40$  MPa. Use the material properties given in the table below to

- (a) Calculate in-plane normal and shear strains;
- (b) Calculate the ratio of shear strain from normal-shear coupling to the total shear strain;
- (c) Calculate the normal stress and strain in fiber direction.



Property	Young's Modulus (Gpa)	Shear Modulus (Gpa)	Poisson's Ratio
Glass	85	35.4	0.2
Epoxy	3.4	1.3	0.38

**Problem 2 (10 points)**

Derive an expression for the in-plane bulk modulus  $\kappa_{23}$  ( $= (\sigma_{22} + \sigma_{33}) / \varepsilon_{kk}$ ) of an unidirectional fiber reinforced composite (fiber direction  $x_1$ ) in terms of the properties of the fiber ( $f$ ) and the matrix ( $m$ ), i.e.,  $\kappa_f$ ,  $E_f$ ,  $\nu_f$ ,  $\kappa_m$ ,  $E_m$ , and  $\nu_m$ , and the volume fraction of the fiber,  $f$ , where  $\kappa$  is bulk modulus,  $E$  is Young's modulus and  $\nu$  is Poisson's ratio.

Use the Lamé solution for a plane-strain composite cylinder consisting of the fiber at the core of radius  $a$  surrounded by the matrix of radius  $b$  subjected to a uniform radial strain  $\varepsilon_0$  at the outer boundary  $r = b$ . The geometrical parameters in the solution are to be eliminated in favor of  $f$ .

- (a) Show that  $\sigma_{rr}|_{r=b} = 2\kappa_{23}\varepsilon_0$ .
- (b) Use (a) to obtain  $\kappa_{23}$ .

**Problem 3 (10 points)**

Consider a thin rectangular plate made of an angle ply ( $\theta$ ) fiber reinforced composite subjected to uniaxial tensile stress  $\sigma$  in the  $x$  direction in the global.

- (a) Compute the transformed compliance matrix  $\bar{S}$  for the angle ply composite in the global coordinate system  $(x, y)$  given the compliance matrix  $S$  in the material coordinate system  $(x_1, x_2)$ . Use engineering constants  $E_1/E_2 = 5$ ,  $G_{12}/E_1 = 0.1$ ,  $\nu_{12} = 0.25$  and  $E_1 = 1$ .
- (b) Compute the state of strain in the plate.
- (c) Determine the ratio of the shear to axial strain for  $\theta = 45^\circ$ . Comment on the deformed shape of the plate by using a simple sketch.
- (d) Determine the non-trivial angle of the ply ( $\theta$ ) for which there is minimal distortion.

